



Multi-Continuum Technology for the Failure of Composites

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- Multi Continuum Technology (MCT) allows for consideration of micro-mechanical effects in macro-sized FEM models
 - Mechanics of fibers and matrix can be considered individually

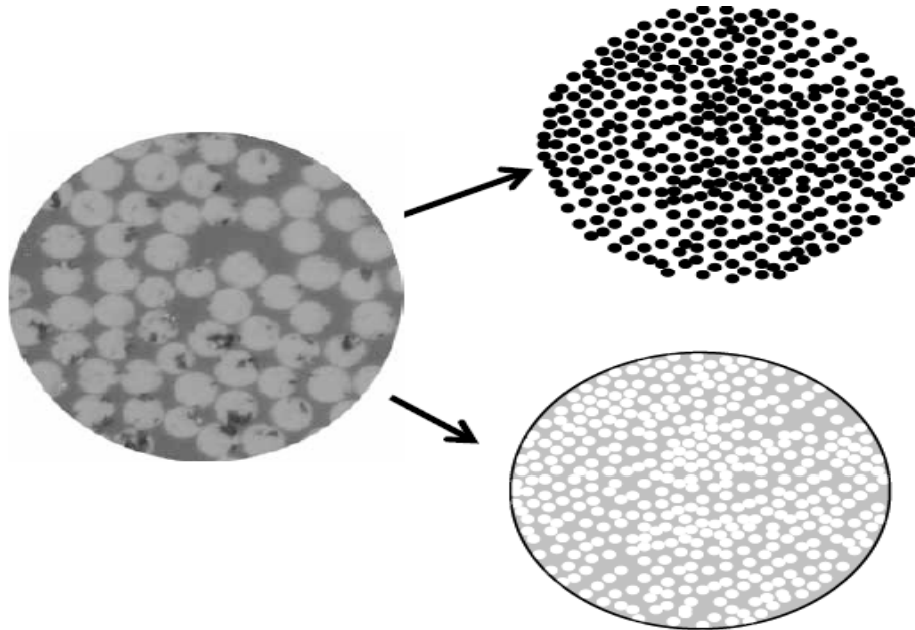
- So what?
 - Much more accurate predictions of composite failure
 - Less need for detail models
 - Better knowledge of material failure point
 - Better and lower safety factors
 - Less material
 - Lower cost and weight
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- MCT Technology developed by Firehole Technology
 - Firehole's Helius product
 - Implemented in NEi Nastran

- Why do we need MCT?
 - Traditional failure theories have been developed based on smeared lamina properties
 - Fibers and matrix are smeared into one
 - Testing and verification needed to develop and tweak models
 - Called the ‘Black Aluminum’ approach
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- So?
- Smearred properties have limited usefulness
 - Failure theories treat composite as a whole
 - Analysis output is for composite, not individual components
 - Limits to what it can accomplish

- MCT separates the fibers and matrices
 - Different stiffness
 - Different failure properties
 - Fiber failure theory for fibers
 - Polymer failure theory for matrix
 - Different Thermal properties
 - Interaction between the two important
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**Better Information
= Better Designs**

FEA Outputs:

$\epsilon_{\text{composite}}, \sigma_{\text{composite}}$

$\epsilon_{\text{fiber}}, \sigma_{\text{fiber}}$

$\epsilon_{\text{matrix}}, \sigma_{\text{matrix}}$

Failure Information:

$FI_{\text{Fiber}}, FI_{\text{Matrix}}$

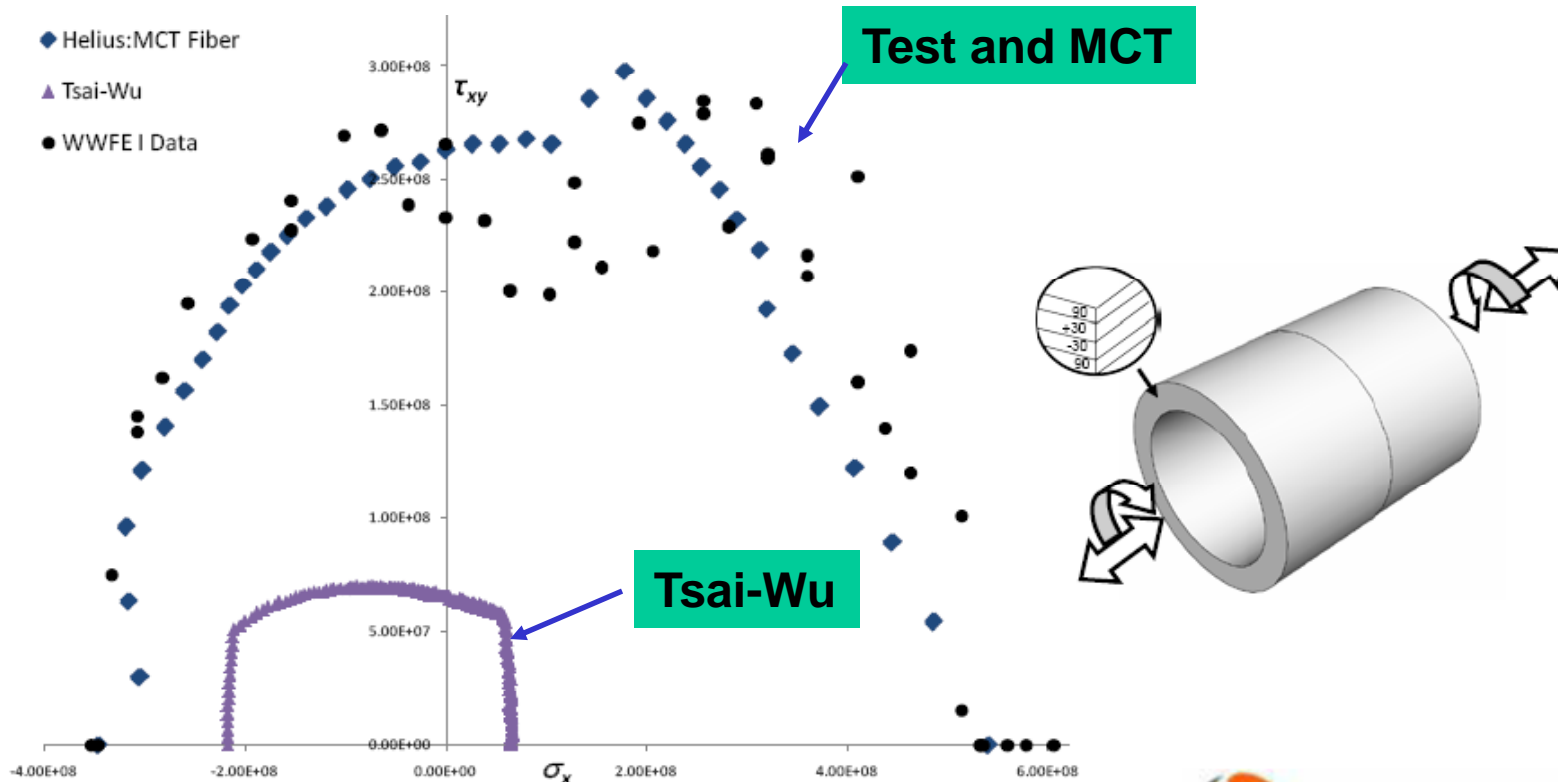
$R_{\text{Fiber}}, R_{\text{Matrix}}$

Exact Failure Mode

- Advantages to this approach
 - Proper failure criteria can be applied to the individual constituents
 - 3D failure theory
 - Off-axis stresses are calculated
 - Interactions can be handled to a degree
 - Can handle unidirectional **and** woven materials

- Hasn't this been done before?
- In a way...
 - Micro-mechanics are usually applied to small details
 - Accurate capture of interaction
 - Smearred properties used for large models
 - Less accurate results
 - **MCT captures the accuracy of the detail, but in a large model**

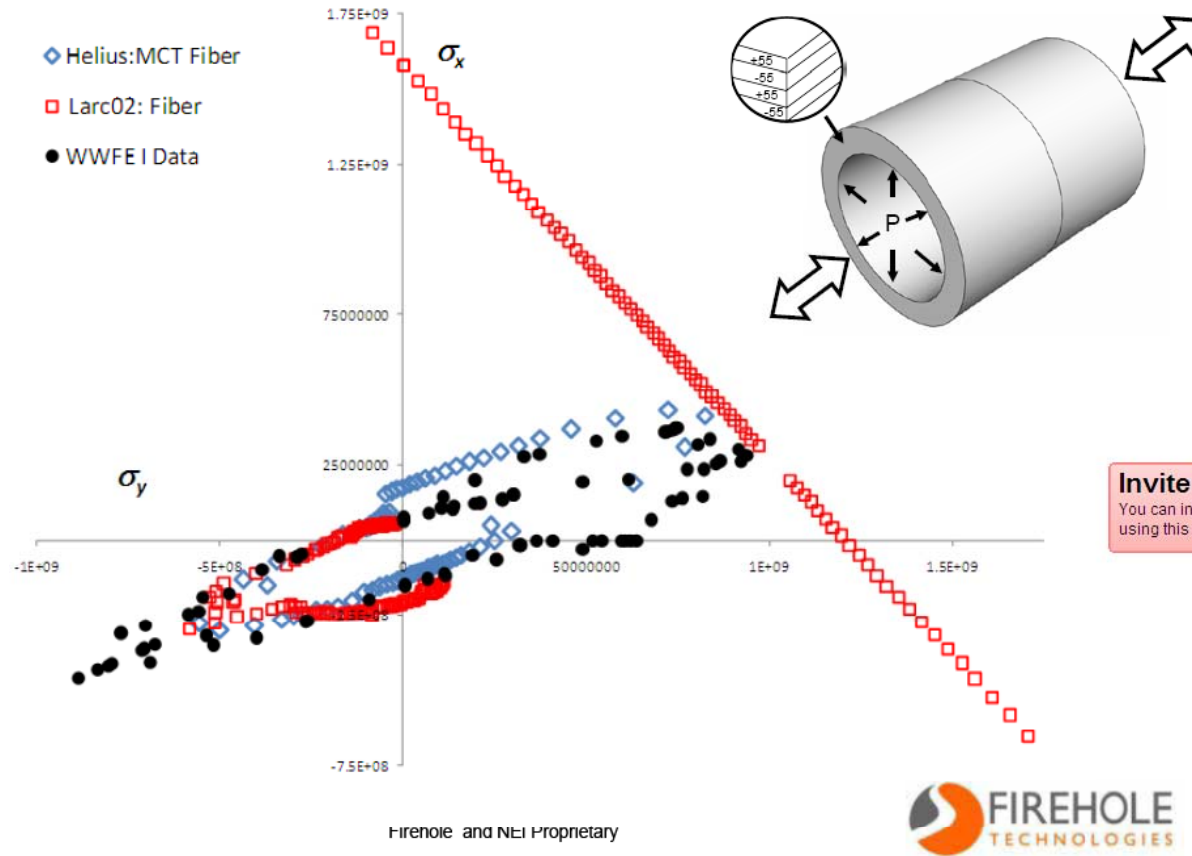
- How do we know this works?
 - The MCT approach has been correlated against many tests and actual failures
 - MCT matches WWFE test results better than most traditional failure theories



Biaxial σ_x : τ_{xy} failure envelope for a [90 / 30]_s glass/epoxy lamina



Biaxial, $\sigma_y:\sigma_x$, failure envelope for a $[\pm 55^\circ]_s$ laminate made from E-glass/Epoxy

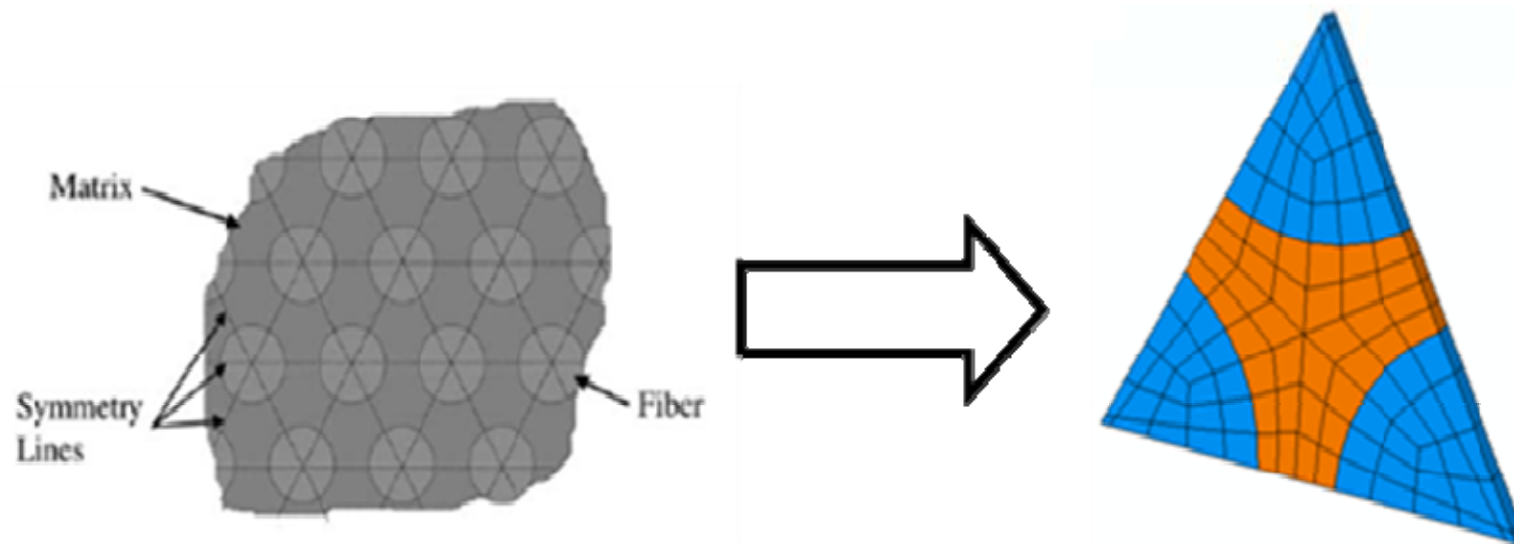


- How does this all work, then?
 - Properties are input for the constituents
 - Fiber properties
 - Matrix Properties
 - Properties are then calculated/optimized for the composite.

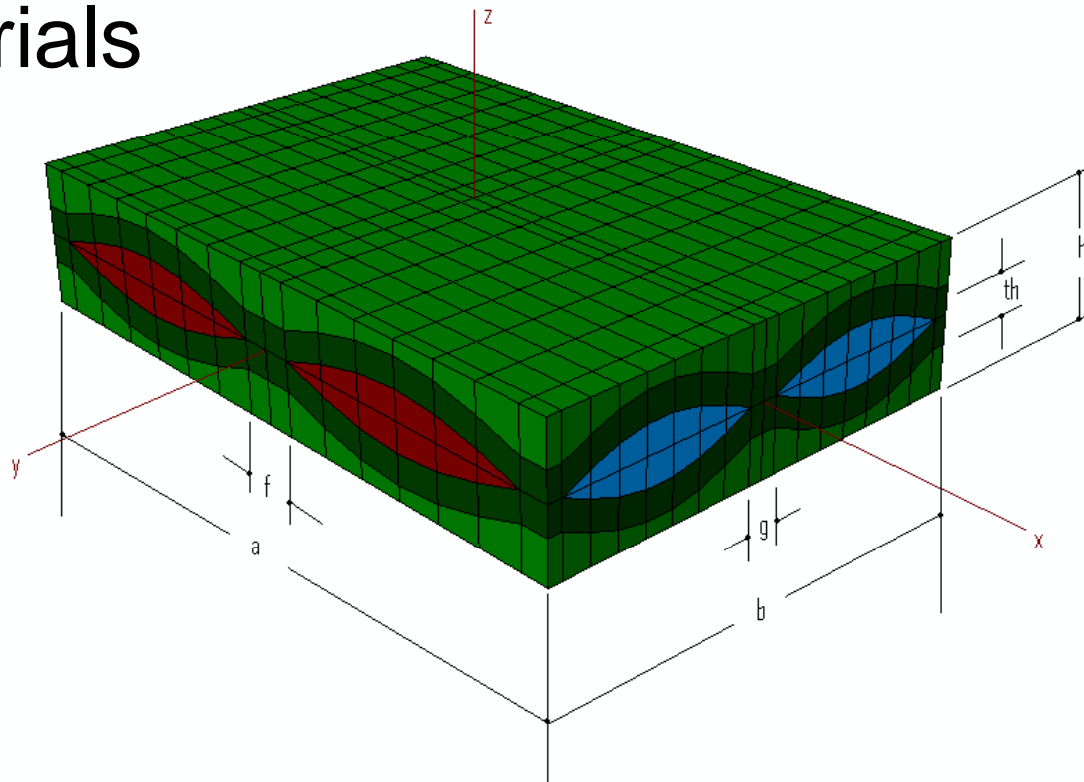
- FEM model is solved with optimized properties, and element strains and stresses are found
- The composite strains and stresses are then decomposed into fiber stresses and fiber strains based in the individual properties and the micro-mechanical models
- Fibers and matrix results can then be handled separately
 - Failure
 - Strength
 - Stiffness

- Ideally, the user inputs material properties for the fibers and matrix
 - Small micromechanical models are constructed and run inside Helius to come up with ‘composite’ properties
 - These ‘optimized’ properties are used for the big FEM model

- Micro-mechanics model for Unidirectional materials



- Micro-mechanics model for woven materials



- The small models are called RVEs “representative volume elements”
- Loads from the model elements are applied to the RVEs
- A detailed 3D picture of the stress state in the individual components can be developed this way
- Reported as fiber and matrix values

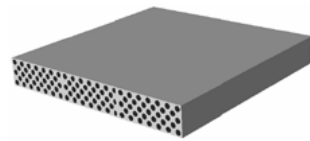
- So what does this buy me?
- A simple thermal example...
 - Consider a block of material subject to a change in temperature

- Without any constraints, this is a zero stress event for an isotropic material
 - But composites are not isotropic!
 - Fibers and matrix have different CTEs
 - Potentially significant stresses in opposite directions in the fiber and matrix
 - Also out of plane!
 - A traditional failure theory would report the composite stress as zero
 - Because it is treated as a isotropic material
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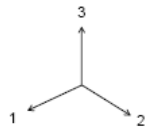
Composite Under Thermal Loading

Composite Stress:

$$\begin{aligned}\sigma_{11} &= 0 \\ \sigma_{22} &= 0 \\ \sigma_{33} &= 0\end{aligned}$$

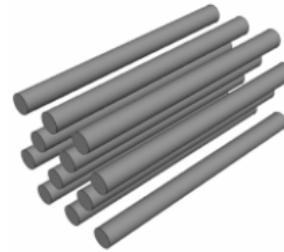


$\Delta T = -216 \text{ }^\circ\text{C}$



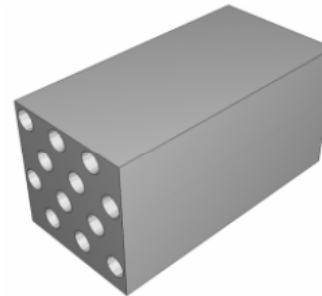
Fiber Stress:

$$\begin{aligned}\sigma_{11f} &= -44.5 \text{ (MPa)} \\ \sigma_{22f} &= -17.25 \text{ (MPa)} \\ \sigma_{33f} &= -17.25 \text{ (MPa)}\end{aligned}$$



Matrix Stress:

$$\begin{aligned}\sigma_{11m} &= 66.75 \text{ (MPa)} \\ \sigma_{22m} &= 25.87 \text{ (Pa)} \\ \sigma_{33m} &= 25.87 \text{ (Pa)}\end{aligned}$$



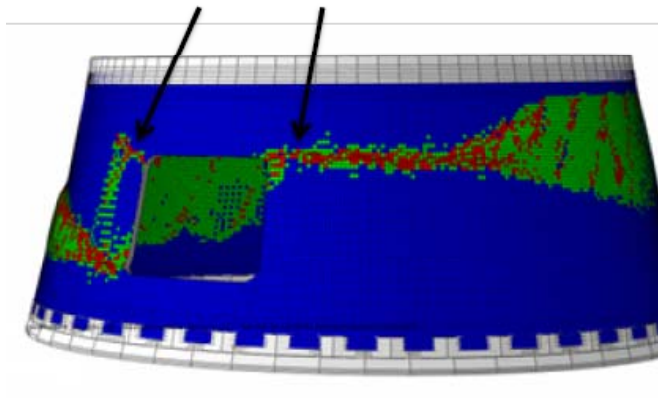
- Structural loading scenarios have similarly unexpected results
 - In-plane loads can generate significant out-of-plane stresses in composites
 - The MCT 3D approach finds these values
 - Delamination may be caused by these loads
 - MCT predicts this better

Results

Helius:MCT Predicted Failure

Red = Fiber Failure

Green = Matrix Failure



✓ **Location**

✓ **2.5%**

- MCT is a much better way to predict the actual limits of composites.
 - Simple to use – relatively few additional properties needed.
 - Better prediction = better design
 - Better design = lower material cost, lower weight
 - Use MCT today!
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